United States of America
Department of Transportation
Federal Aviation Administration

PROPELLER
Type Certificate

IMPORT

Number: P00126IB

This certificate issued to: Electravia SAS
t/a Electravia Hélices E-Props

Certifies that the type design for the following product with the operating limitations and conditions therefore as specified in the Federal Aviation Regulations and the Type Certificate Data Sheet meets the airworthiness requirements of Part 35 of the Federal Aviation Administration Title 14 CFR .

Model: EPGU3

This certificate, and the Type Certificate Data Sheet which is a part hereof, shall remain in effect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of Application: 08/06/2026 Date Reissued: _____

Date of Issuance: 11/24/2026 Date Amended: _____

By Direction of the Administrator

Signature: TIMOTHY J HADSALL Digitally signed by TIMOTHY J HADSALL Date: 2026.11.24 10:44:51 -05'00'

Title: Timothy Hadsall
Manager, International Validation Branch
Compliance & Airworthiness Division
Aircraft Certification Service

Any alteration of this certificate and/or the Type Certificate Data Sheet is punishable by a fine not exceeding $1,000, or imprisonment not exceeding 3 years, or both. This certificate may be transferred or made available to third persons by licensing agreements in accordance with Title 14 of the Code of Federal Regulations, part 21, section 21.47 (14 CFR 21.47). A transfer must be endorsed as provided on the reverse hereof. A Type Certificate holder who allows a person to use the Type Certificate to manufacture a new aircraft, aircraft engine, or propeller must provide that person with a written licensing agreement acceptable to the FAA. (Ref. 14 CFR 21.55).

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Transfer Endorsement

Before transferring a certificate, the person making the transfer must notify in writing the appropriate aircraft certification office. The notification must include the type certificate number, name and address of the transferee, and anticipated date of the transfer (Ref. 14 CFR 21.47).

To (Name and address)

By: _____ Date: _____

Holder: _____

Signature: _____

Title: _____

To (Name and address)

By: _____ Date: _____

Holder: _____

Signature: _____

Title: _____

To (Name and address)

By: _____ Date: _____

Holder: _____

Signature:

Title:

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U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION TYPE CERTIFICATE DATA SHEET P00126IB	TCDS NUMBER: P00126IB REVISION: 0 ELECTRAVIA HELICES E-PROPS MODEL: EPGU3 DATE: [November 24, 2026]

Propellers of models described herein conforming with this data sheet (which is part of this Type Certificate No. P00126IB) and other approved data on file with the Federal Aviation Administration, meet the minimum standards for use in certified aircraft in accordance with pertinent aircraft data sheets and applicable portions of the Federal Aviation Regulations provided they are installed, operated and maintained as prescribed by approved manufacturer's manual and other approved instructions.

TYPE CERTIFICATE HOLDER	Electravia SAS t/a Electravia Hélices E-Props 195 Route de l'Aviation 04200 Vaumeilh France
TYPE	Ground adjustable pitch propeller
ENGINE MOUNTING	See Note 1 of this TCDS
HUB MATERIAL	Carbon fiber reinforced composite
BLADE MATERIAL	Carbon fiber reinforced composite, foam core, leading edge erosion protection
HUBS:	See Note 1 of this TCDS
NUMBER OF BLADES	3 (three)
DESIGN SERIES	EPGU3

HUB-TYPE EPGU3 See Note 1	BLADES See Notes 2 & 6	MAXIMUM CONTINUOUS		TAKE OFF		NOMINAL DIAMETER Max		Min		Pitch Angle *)		APPROXI-MATE WEIGHT	
		HP(kW)	RPM	HP(kW)	RPM	inch	(cm)	inch	(cm)	Min	Max	lbs.	(kg)
EPGU3-( )A	310	93 (69)	2265	99 (73.5)	2388	63	(160)	61	(155)	N/A		4.4	(2.0)
	312					69	(175)	65	(165)			4.6	(2.1)
	314					75	(190)	71	(180)			4.9	(2.2)

*) Not applicable: ground adjustable pitch propeller, no pitch angle range is defined (see Note 3 of this TCDS).

CERTIFICATION BASIS:	Pursuant to 14 CFR 21.29(a)(1)(ii), the Type Certificate was issued in validation of the European Union Aviation Safety Agency (EASA) Certification Standards CS-22, Amendment 3 Subpart J dated 15 September 2021, which was found to provide a level of safety equivalent to that provided by 14 CFR Part 35 effective February 1, 1965, as amended by 35-1 through 35-10. Application date to EASA was 28 January 2022. EASA issued Type Certificate P.512 for the EPGU3 series propeller on 02 March 2023.
TC (IMPORT) NO.	P00126IB
TC APPLICATION DATE:	[August 06, 2026]
TC ISSUED	[November 24, 2026]
PRODUCTION BASIS:	EASA Production Organization Approval: FR.21G.0345
IMPORT REQUIREMENTS:	To be considered eligible for installation on U.S. registered aircraft, each propeller imported into the United States must be accompanied by a certificate of airworthiness for export or certifying statement endorsed by the exporting cognizant civil airworthiness authority. Include the following language in the certifying statement: (1) This propeller conforms to its United States type design (Type Certificate Number P00126IB) and is in a condition for safe operation; and (2) This propeller has been subjected by the manufacturer to a final operational check and is in a proper state of airworthiness. Reference 14 CFR 21.500, which provides for the airworthiness acceptance of engines or propellers manufactured outside the U.S. for which a U.S. type certificate has been issued. Additional guidance is contained in FAA Advisory Circular 21-23, Airworthiness Certification of Civil Aircraft, Engines, Propellers, and Related Products, Imported into the United States.

NOTES

NOTE 1: HUB MODEL DESIGNATION:

EP	G	U	3	-	( )	A
1	2	3	4		5	6

1 Electravia Hélices E-Props

2 Hub type: G = Ground adjustable pitch propeller

3 Identification of hub model

4 Number of blades: 3

5 Hub extension: length in mm, 2 or 3 digits
- 39 to 179 mm, every 5 mm; other dimensions on request

6 Letter code for flange type:
- A = 6x 13 mm holes on a 4" (101.6 mm) bolt circle diameter

NOTE 2: BLADE MODEL DESIGNATION:

( )	312	a	-	175	A
1	2	3		4	5

1 Letter code for direction of rotation:
- blank = right-hand (viewed in flight direction)

2 Identification of blade design (3 digits): 310, 312, 314

3 Blade revision letter

4 Diameter in cm

5 Letter code for blade finishing:
- A = standard polished surface
- F = paint on full blade
- P = paint on tips

NOTES CONTINUED

NOTE 3:	Pitch Control:	Ground adjustable pitch propeller: the blade pitch angle is set on the ground only, the propeller incorporates no pitch change mechanism and no control system.
NOTE 4:	Feathering and Reversing:	(a) Feathering: Model incorporates no feathering feature. (b) Reversing: Model incorporates no reversing feature.
NOTE 5:	Right & Left Hand Models:	Only right-hand rotation (viewed in flight direction) is approved.
NOTE 6:	Interchangeability:	See NOTE 1 and NOTE 2.
NOTE 7:	Accessories:	(a) Propeller Spinners: An optional spinner (carbon fiber dome with bulkhead attached to the propeller hub) is available in different shapes and diameters. (b) Propeller Governors: Not applicable. The spinner is not included in the certified Type Design. Related propeller equipment must be approved as part of the aircraft installation regardless of manufacture.
NOTE 8:	Shank fairings:	Not applicable
NOTE 9:	Special limits:	The propeller is approved for geared engines only; it must not be operated on direct drive engines. The suitability of a propeller for a given aircraft/engine combination must be demonstrated within the scope of the type certification of the aircraft.
NOTE 10		The propeller installation must be approved as part of the aircraft type certificate to demonstrate compliance with the applicable aircraft airworthiness standards.
NOTE 11	Special limits:	The FAA approved Airworthiness Limitations Section of the Instructions for Continued Airworthiness is published in the applicable "Installation and Maintenance Manual" document number EP-IMM-001, chapter 8 "Airworthiness Limitations Section". This ALS section specifies a mandatory periodic inspection interval of 100 flight hours or 12 calendar months (whichever comes first), a Time Between Overhaul (TBO) of 2,000 flight hours or 84 calendar months (whichever comes first), and a propeller life limit of 4,000 flight hours. All EPGU3 propellers operating limitations are defined in E-Props Installation and Maintenance Manual No. EP-IMM-001 Propeller Maintenance, or Overhaul, shall be accomplished in accordance with E-Props Repair and Overhaul Manual No. EP-ROM-001 and Illustrated Parts List No. EP-IPL-001, or other methods approved or accepted by the FAA. The propeller CMCB must evaluate the propeller installation for each new aircraft installation to assess possible changes in airworthiness limitations.
NOTE 12:	Special notes:	The propeller is eligible for installation on powered sailplanes, very light aeroplanes and aircraft which can accept a propeller certified according to 14 CFR Part 35.
*NOTE 13:	Service Information:	Each of the documents listed below must state that it is approved by the European Union Aviation Safety Agency (EASA). Any such documents are accepted by the FAA and are considered FAA approved. - Service bulletins, - Structural repair manuals, - Vendor manuals, - Aircraft flight manuals, and - Overhaul and maintenance manuals.

-- END --